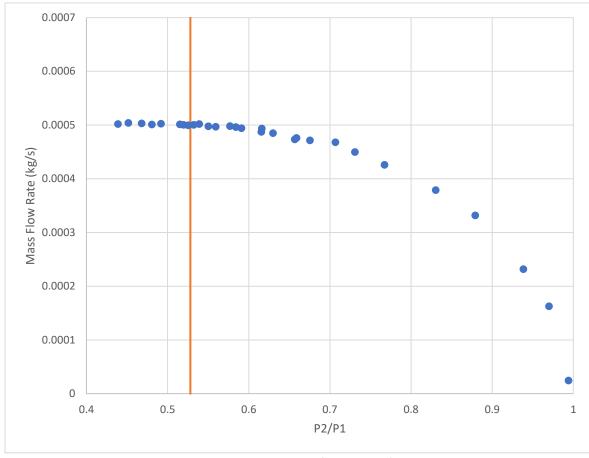
## Key

a) The flow in this system is adiabatic so we can use the following equation:

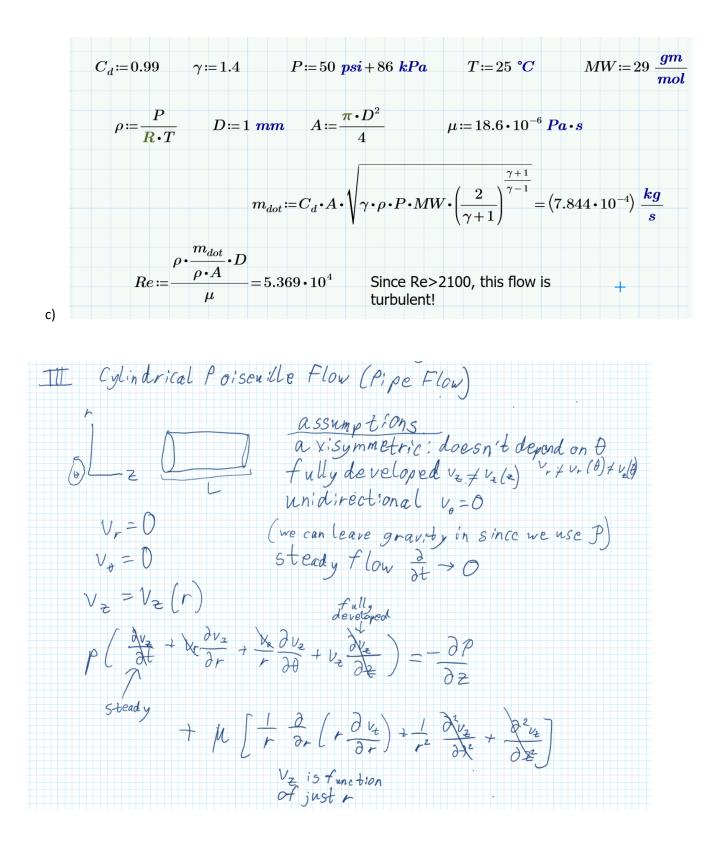
$$1 + \left(\frac{\gamma - 1}{2}\right) \mathrm{Ma}^2 = \frac{T_0}{T} = \left(\frac{P_0}{P}\right)^{\frac{\gamma - 1}{\gamma}}$$

Where  $P_0$  is the upstream pressure and P is the downstream pressure. By setting the Mach number equal to 1 (since we want the flow to be choked) and using a heat capacity ratio of 1.4, the pressure ratio (P/P<sub>0</sub>) is found to be 0.528.



The orange line represents the pressure ratio solved for in part a). Below this pressure ratio, the mass flow rate in constant. This means that, at choked flow, we have the maximum possible mass flow rate that can occur through this orifice. Having choked flow could be useful in reactors that need a guaranteed high amount of cooling water.

b)



 $O = \frac{-\partial P}{\partial z} + \mu \frac{1}{F} \frac{\partial}{\partial r} \left( r \frac{\partial v_{z}}{\partial r} \right)$ Boundary conditions  $V_z(R) = 0$  from no slip assumption Vz(0) C Symmetry condition

d) Yes, we can solve for the velocity profile with the given conditions and simplifications.